## AMENDMENTS TO THE CLAIMS

This listing of claims will replace all prior versions, and listings, of claims in the application.

## **Listing of Claims:**

Claim 1 (Currently Amended): Method of controlling a landing guide path of an aircraft including a fuselage, two wings protruding laterally from opposite sides of said fuselage, and a tail unit extending aft from said fuselage, the method comprising:

adjusting control surfaces on said wings and said tail unit; and

rotating a <u>respective</u> portion of an <u>a respective</u> aerodynamic component <u>connected to</u> a <u>respective outboard wingtip of each of said two wings</u> to increase a drag without influencing a lift.

Claim 2 (Currently Amended): The method according to claim 1, wherein the aerodynamic component comprises comprises a main body configured to be connected to a wing one of said two wings and a control member connected to the main body, and said rotating comprises rotating a portion of the control member.

Claim 3 (Currently Amended): The method according to claim 2, wherein the control member comprises a fixed member connected to the main body and a pivotable member connected to the fixed member, said pivotable member being said portion, and said rotating comprises rotating the pivotable member.

Claim 4 (Currently Amended): The method according to claim 3, wherein the control member further comprises a hinge member disposed between the fixed member and the

pivotable member, and <u>said</u> rotating comprises rotating the pivotable member relative to the fixed member via the hinge member.

Claim 5 (Original): The method according to claim 4, wherein the pivotable member comprises a delta shape.

Claim 6 (Currently Amended): The method according to claim 4, wherein <u>said</u> rotating comprises rotating the pivotable member about an axis perpendicular to a major plane of the <u>wing said one of said two wings</u>.

Claim 7 (Currently Amended): The method according to claim 4, wherein <u>said</u> rotating comprises rotating the pivotable member at least one of inwardly and outwardly.

Claim 8 (Currently Amended): The method according to claim 4, wherein <u>said</u> rotating comprises rotating the pivotable member both of inwardly and outwardly.

Claim 9 (Currently Amended): Method of steepening a landing guide path of an aircraft including a fuselage, two wings protruding laterally from opposite sides of said fuselage, and a tail unit extending aft from said fuselage, the method comprising:

adjusting control surfaces on said wings and said tail unit; and
rotating a respective portion of an a respective aerodynamic component connected to
an outboard wingtip of each of said two wings to increase a drag without influencing a lift.

Claim 10 (Currently Amended): The method according to claim 9, wherein the aerodynamic component comprises a main body configured to be connected to a

wing one of said two wings and a control member connected to the main body, and said rotating comprises rotating a portion of the control member.

Claim 11 (Currently Amended): The method according to claim 10, wherein the control member comprises a fixed member connected to the main body and a pivotable member connected to the fixed member, said pivotable member being said portion, and said rotating comprises rotating the pivotable member.

Claim 12 (Currently Amended): The method according to claim 11, wherein the control member further comprises a hinge member disposed between the fixed member and the pivotable member, and <u>said</u> rotating comprises rotating the pivotable member relative to the fixed member via the hinge member.

Claim 13 (Currently Amended): Method of controlling a landing guide path of an aircraft including a fuselage, two wings protruding laterally from opposite sides of said fuselage, and a tail unit extending aft from said fuselage, each of said wings including an a respective aerodynamic component having a main portion and a control portion, the main portion connected to a wing of the aircraft, and the control portion including a fixed member connected to a pivotable member, the method comprising:

adjusting control surfaces on said wings and said tail unit; and rotating the pivotable member to increase a drag without influencing a lift.

Claim 14 (Currently Amended): The method according to claim 13, wherein <u>said</u> rotating comprises rotating the pivotable member perpendicular to a major plane of the wing one of said wings.

Claim 15 (Currently Amended): The method according to claim 14, wherein <u>said</u> rotating comprises rotating the pivotable member in at least one of inwardly and outwardly.

Claim 16 (Currently Amended): The method according to claim 15, wherein <u>said</u> rotating comprises rotating the pivotable member both inwardly and outwardly.

Claim 17 (New): The method according to claim 1, wherein said rotating is performed on the respective portions of the respective aerodynamic components on both of said wings in a manner synchronized with one another.

Claim 18 (New): The method according to claim 1, wherein said rotating is performed on the respective portions of the respective aerodynamic components on both of said wings in a manner symmetrical with one another.

Claim 19 (New): The method according to claim 1, wherein said rotating is performed independently for the respective portions of the respective aerodynamic components on said two wings.

Claim 20 (New): The method according to claim 1, wherein said rotating is performed independently of said adjusting of said control surfaces.